UNCLASSIFIED

AD 296 254

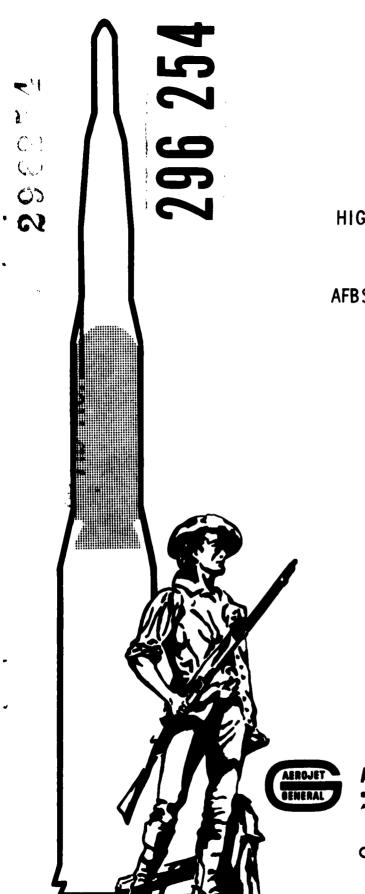
Reproduced by the

ARMED SERVICES TECHNICAL INFORMATION AGENCY
ARLINGTON HALL STATION
ARLINGTON 12, VIRGINIA



UNCLASSIFIED

NOTICE: When government or other drawings, specifications or other data are used for any purpose other than in connection with a definitely related government procurement operation, the U. S. Government thereby incurs no responsibility, nor any obligation whatsoever; and the fact that the Government may have formulated, furnished, or in any way supplied the said drawings, specifications, or other data is not to be regarded by implication or otherwise as in any manner licensing the holder or any other person or corporation, or conveying any rights or permission to manufacture, use or sell any patented invention that may in any way be related thereto.



19 December 1962

Report No. 0162-01TN-62-18

HIGH- AND LOW-TEMPERATURE TESTS
OF THE AEROJET UTILITY VAN

AFBSD Technical Note BSD-TDR-62-331

Contract No. AF 33(600)-36610

AEROJET-GENERAL CORPORATION

COPY NO. 15

W. O. 0380

¥

Report No. 0162-01TN-62-18

HIGH AND LOW-TEMPERATURE TESTS OF THE AEROJET UTILITY VAN

AFBSD Technical Note BSD-TDR-62-331

Contract No. AF 33(600)-36610

Prepared for

HQ., AIR FORCE BALLISTIC SYSTEMS DIVISION
AIR FORCE SYSTEMS COMMAND
AIR FORCE UNIT POST OFFICE
Los Angeles 45, California
ATTN: Tech Data Center

Approved by:

C. C. Coaway, Manager Minuteman Program



SOLID ROCKET PLANT SACRAMENTO, CALIFORNIA

A SUBSIDIARY OF THE GENERAL TIRE & RUBBER COMPANY

PREFACE

Acknowledgment is made to the following persons for the preparation of this report: Jack D. Sohl, Environmental Program Engineer; C. M. McCarty, Environmental Testing Department; Max Halebsky, Minuteman Environmental Program; and D. P. Campbell, Technical Editor.



ABSTRACT

The Aerojet-General Utility van has successfully completed high- and low-temperature tests to verify the acceptability of the van as a transport vehicle for second-stage Minuteman operational motors. A Minuteman motor cast with propellant simulant was used in the tests. While the exterior of the van was subjected to high and low temperatures, ranging from +140 to -35°F, the surface of the motor was maintained within the specified temperature range of 80+20°F.



TABLE OF CONTENTS

			Page No.
I.	Intr	oduction	1
II.	Tec	hnical Discussion	1
	A.	Test Units	i
	B.	Instrumentation	2
	C.	Test Conditions	2
	D.	Test Results	3
III.	Con	clusions	5



FIGURE LIST

	Figure No.
Aerojet Utility Van Positioned in Engine Test Cell Climatic Laboratory, Air Proving Ground Center, Eglin AFB, Florida	1
Thermo-King Temperature-Control Unit Installed on Aerojet Utility Van	2
Locations and Designations of Thermocouples	3
Minneapolis-Honeywell Brown Print-out Recorders and Standard Reference Junction	4
Calibration Plot of Print-out Recorders	5
Instrumentation at Aft End of Motor	6
Instrumentation at Forward End of Motor	7
Instrumentation at Left Side of Motor	8
Instrumentation at Right Side of Motor	9
Thermocouple Termination at Aft End of Utility Van	10
Low-Temperature Test Data (Plot of Thermocouples A2, B6, and C2)	11
Low-Temperature Test Data (Plot of Thermocouples A5, B4, and C5)	12
High-Temperature Test Data (Plot of Thermocouples A2, B6, and C2)	13
High-Temperature Test Data (Plot of Thermocouples A5, B4, and C5	14
APPENDIXES	
Minuteman Engineering and Inspection Summary	A
Temperature Surveillance Chart, High- and Low-Temperature Tests of Aerojet Utility Van	В
Chronological List of Events, High- and Low-Temperature Tests of Aerojet Utility Van	С



I. INTRODUCTION

To verify the acceptability of the Aerojet Utility van, manufactured by the Utility Trailer Manufacturing Co., as a transport vehicle for Minuteman secondstage operational motors, the van was subjected to high- and low-temperature tests at Eglin Air Force Base, Florida. An inert second-stage motor was housed in the van. These tests were made in conjunction with structural verification tests of the van following completion of the Aerojet Utility Van Certification Program.

II. TECHNICAL DISCUSSION

A. TEST UNITS

1. Test Facility

The climatic testing was done in the Engine Test Cell, Climatic Laboratory, Air Proving Ground Center, Eglin AFB, Florida. The Aerojet Utility van installed in the cell is shown in Figure 1.

2. Aerojet Utility Van

The Aerojet Utility van is a trailer designed to house a Minute-man second-stage operational motor while in transit by ground or air transport. The test van, model VR-1152A, SN 37290 (AF 348873), had been modified in accordance with Aerojet drawing T-492674.

3. Heating and Cooling Unit

The interior temperature of the van was regulated by a Thermo-King temperature-control unit, Model K30A. The unit, installed on the exterior forward wall of the van (Figure 2), provided heat from three heater strips powered by



II, A, Test Units (cont.)

220 vac. Cooling was provided by the refrigeration section of the unit. In accordance with standard summer practice, the louvered top cover and shutters of the unit had been removed.

4. Test Motor

The test motor was Minuteman second-stage motor 44TE-1, (SN 519592). The motor had been cast with propellant simulant.

B. INSTRUMENTATION

The motor and interior of the van was instrumented with 26 iron-constantan thermocouples (Figure 3) to determine motor-surface and van-interior temperature gradients. All temperature data were continously recorded on two Minneapolis-Honeywell Brown print-out recorders, Model No. RY153X82-C-II-III (13), serial numbers 322005 and 314315 (shown in Figure 4 with standard reference junction). Calibration input voltage for each 10°F from -50 to +200°F was applied to each channel by a Leeds-Northrup potentiometer, Model K-2. A plot of this calibration is shown in Figure 5. A continuous +150°F reference point was provided on each recorder by a standard reference junction manufactured by Pace Engineering Company. Figures 6 through 9 show the instrumentation of the motor. Figure 10 shows the thermocouple termination at the aft end of the Utility van.

C. TEST CONDITIONS

In accordance with Aerojet Test Plan 752, Revision A, the Utility van was subjected to high- and low-temperature tests. The low extreme was -35 ± 5 °F, and the high extreme was +140 + 5°F, including 15°F as radiant energy of the sun.

1. Low-Temperature Test

With the motor surface temperature at 80 ± 5 °F, the temperature of the test cell was reduced to -35°F and held until 4 hr after the surface temperature



II, C, Test Conditions (cont.)

of the motor had stabilized. Then the Thermo-King unit was switched off and temperatures were monitored for another 4 hr while the test-cell temperature remained at -35°F.

2. High-Temperature Test

After the motor surface temperature returned to and was stabilized at 80°F, the temperature of the conditioning cell was increased to +140°F and held for 4 hr. The Thermo-King unit was then switched off. While the temperature of the test cell was maintained at +140°F, temperature data were recorded for another 4 hr.

D. TEST RESULTS

1. General

Instrumentation of the Utility van and inert motor was completed at Aerojet-General, Sacramento prior to shipment. The tractor-van, traveling over conventional highway routes, arrived at Eglin AFB, Florida on 30 July 1962.

No major operational difficulties or malfunctions of equipment or instruments occurred during the tests. The temperature of the test cell was maintained within the specified tolerances. Air temperatures at the forward and aft areas of the van were recorded throughout the tests. A buildup of ice and frost near the top seal at the aft doors of the van indicated a source of heat loss during the low-temperature test.

All test objectives were accomplished satisfactorily. The tests were witnessed by an inspector representing the Air Force Systems Command, Field Test Office. Inspection and documentation were accomplished in accordance with Minuteman Engineering and Inspection Summary (MEIS) HT-62, as given in Appendix A.



II, D, Test Results (cont.)

Appendix B lists recorded temperatures for the low- and high-temperature tests. Test events are listed chronologically in Appendix C.

2. Low-Temperature Test

Calibration of instruments and preparation of the test cell were completed on 8 August. On 9 August, the Utility van was positioned in the cell and installation of the exhaust-gas duct, servicing of the Thermo-King unit, and thermocouple termination were completed.

At 1030 hours on 9 August, temperature conditioning of the cell was started. The required low temperature of 35 ±5°F for the interior of the cell was recorded at 1500 hours and temperature stabilization of the motor began. The test proceeded without incident except for a period of 1.5 hr during which the operation of the Thermo-King unit was intermittent. A defective circuit in post-to-fuse connection was discovered, and corrective action was taken. At 0700 hours on 10 August, 16 hr after the test began, the surface temperature of the motor stabilized at +60°F. At 1100 hours, the Thermo-King unit was switched off. The average van-interior and motor-surface temperatures at this time were 59 and 60°F, respectively. After 4 hr without heating, the temperature recordings of the van interior ranged from 32 to 38°F, and the motor-surface temperature ranged from 45 to 49°F. An attempt to restart the Thermo-King unit on conclusion of the test was unsuccessful. At 1505 hours, the cell temperature was returned to +80°F and the van doors were opened. Figures 11 and 12 show the recorded low-temperature data in graph form.

3. High-Temperature Tests

At 0745 hours on 13 August, temperature conditioning of the cell was started. The required high temperature of +140 ±5°F was recorded at 0900 hours. Since van-interior temperatures ranged from 74 to 77°F, the thermostat setting of the Thermo-King unit was reduced from 80 to 75°F to



II, D, Test Results (cont.)

prevent commencement of the heating cycle. The Thermo-King unit was manually switched on at 0915 hours for a functional check of the unit. This resulted in a drop of the van-interior temperature between 0930 and 1000 hours. The 4-hr test period was completed at 1305 hours when the motor-surface temperature was +87°F and the average van-interior temperature was +89°F. The Thermo-King unit was then switched off and the unit remained inoperative for 4 hr. The highest motor-surface temperature recorded during this period was +94°F, and the highest van-interior temperature was +104°F. At 1706 hr, the Thermo-King unit was operated for 10 min. The van-interior temperature was reduced to an average of +90°F, and the motor-surface temperature to +88°F. Figures 13 and 14 show the high-temperature data in graph form.

III. CONCLUSIONS

During the low-temperature tests, the motor-surface equilibrium temperature was determined to be +60°F. During the high-temperature test, the highest motor-surface temperature recorded was +87°F. Therefore, on the basis of the acceptance criteria as given in Space Technology Laboratories, Inc., document 62-9731.3-523, dated 22 March 1962, the Aerojet Utility van is acceptable as a transport vehicle for Minuteman second-stage operational motors.

Data obtained when the Thermo-King temperature-control unit was not operating indicate that an actual malfunction of the unit (occurring under temperature conditions similar to those imposed during the tests) would not adversely affect a Minuteman motor. If the motor-surface temperature was at +80°F at time of unit failure, the motor-surface temperature would not increase beyond 100°F during the first 10 hr of exposure to the high-temperature extreme (+140°F). Similarly, during 10 hr of exposure to the low-temperature extreme (-35°F), the motor-surface temperature would not fall below +40°F.



III, Conclusions (cont.)

During the operation of the Thermo-King unit, the temperature within the van remained steady. A variance of 2°F was the greatest recorded. Data obtained indicate the mode of temperature stratification during periods when the Thermo-King unit is not operating. The temperature of the air layer at the ceiling of the van is 6°F higher than the air layer at the floor of the van. There is no apparent difference in air temperature in the fore and aft sections of the van.



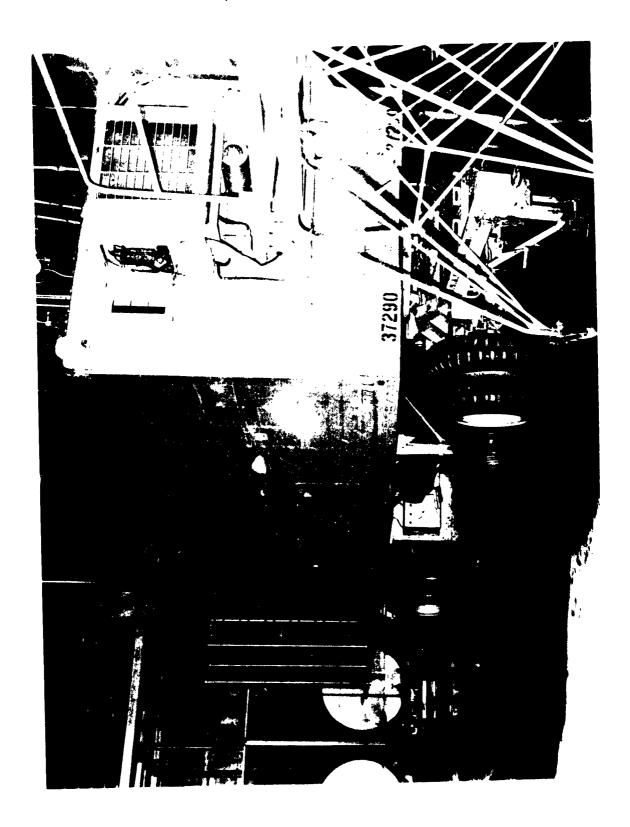
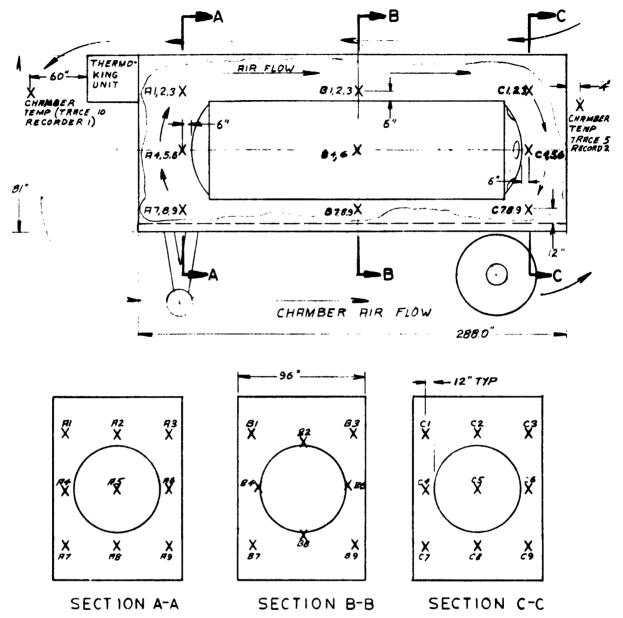


Figure 1



Thermo-King Temperature-Control Unit Installed on Aerojet Utility Van (View of Right-Forward Corner of Van)



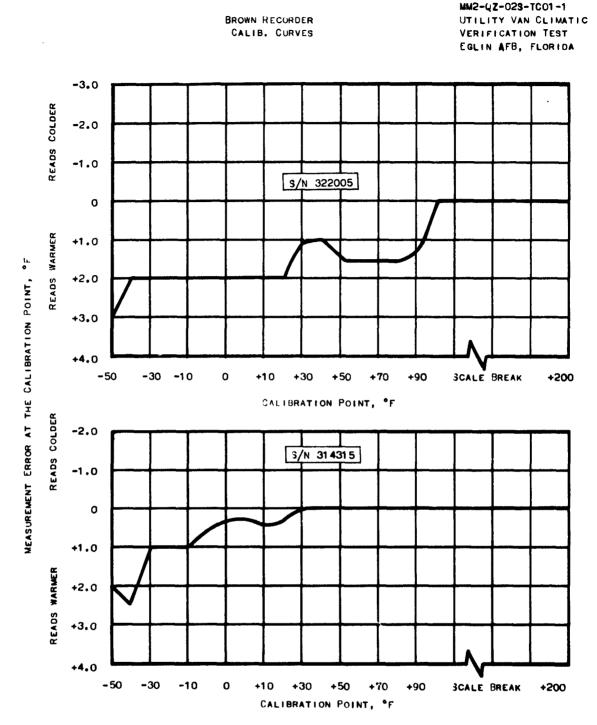
Note:

1. All free air thermocouples are secured to prevent swinging caused by air flow. 2. Thermocouples 32, $3h_1$, 36 and 33 are measuring meter surface temperature.

Locations and Designations of Thermocouples



Figure 4



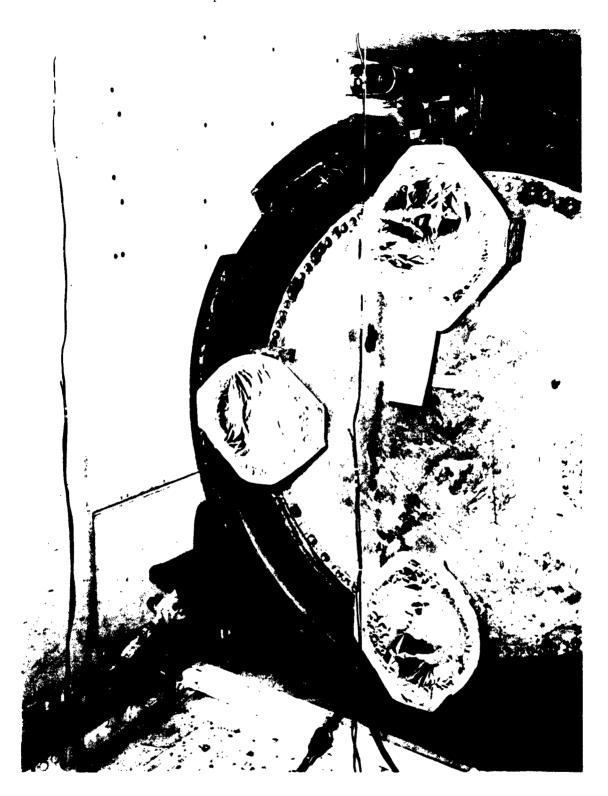
CALIBRATION INPUT VOLTAGE AT EACH POINT SUPPLIED BY LEEDS AND NORTHRUP K-2 POTENTIOMETER.

CALIBRATIONS OBTAINED ON 8 AUGUST 1962.

ORIGINAL CALIBRATION RECORDS ON FILE WITH AFSC FIELD TEST SECTION, EGLIN AFB, FLORIDA.

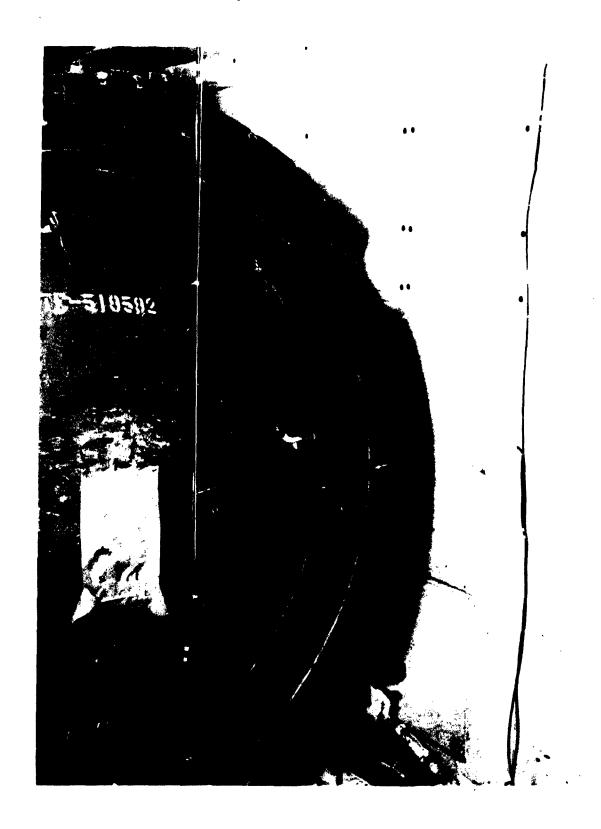
Calibration Plot of Print-Out Recorders

Figure 5



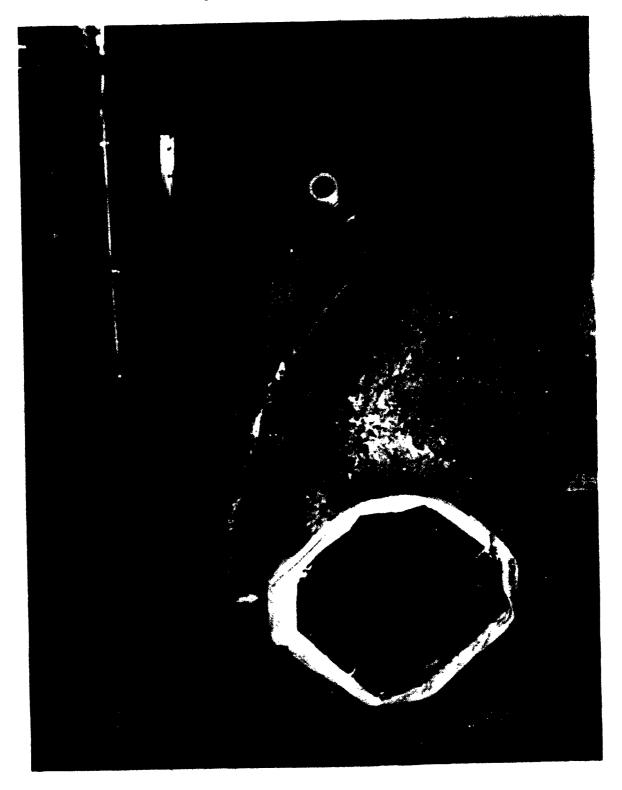
Instrumentation at Aft End of Motor

Figure 6



Instrumentation at Forward End of Motor

Figure 7



Instrumentation at Left Side of Motor

Figure 8



Instrumentation at Right Side of Motor

Figure 9

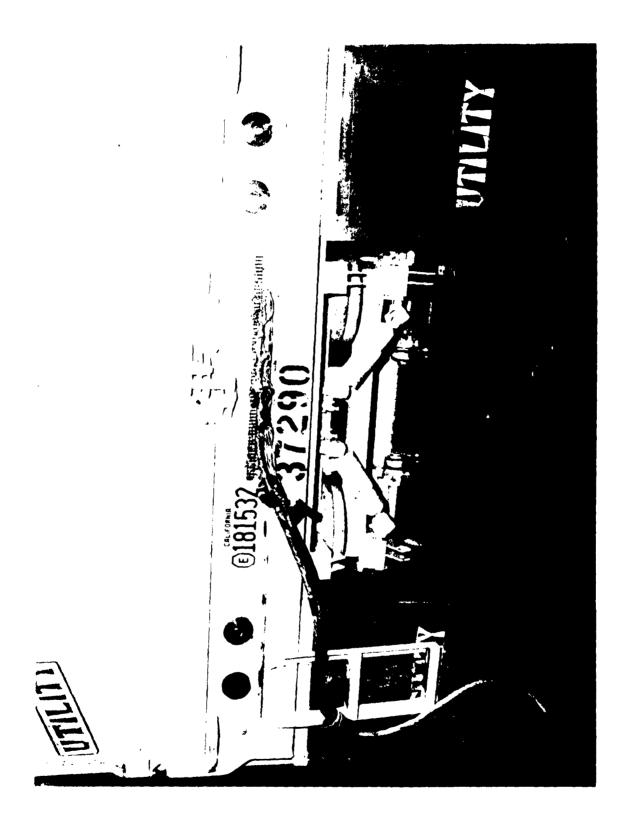
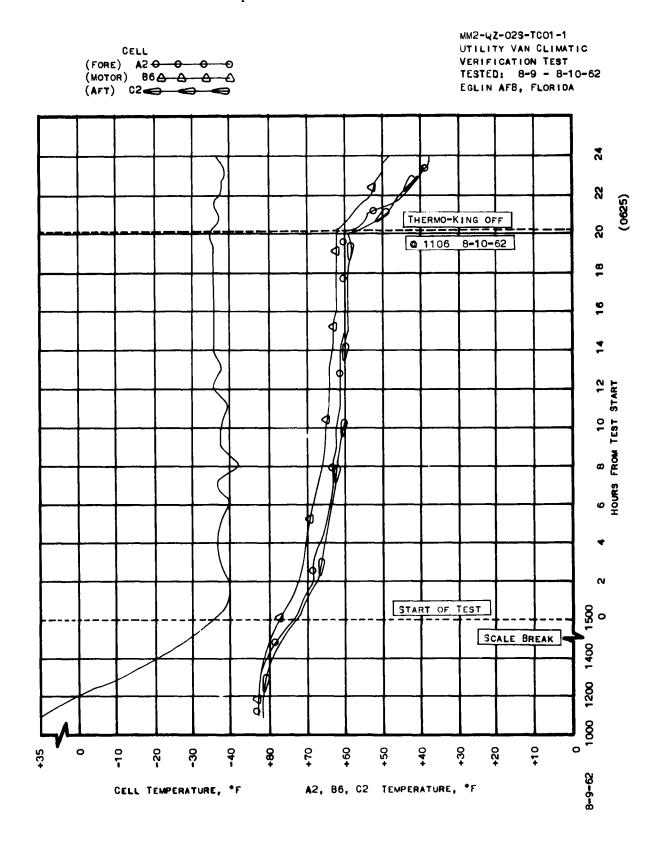
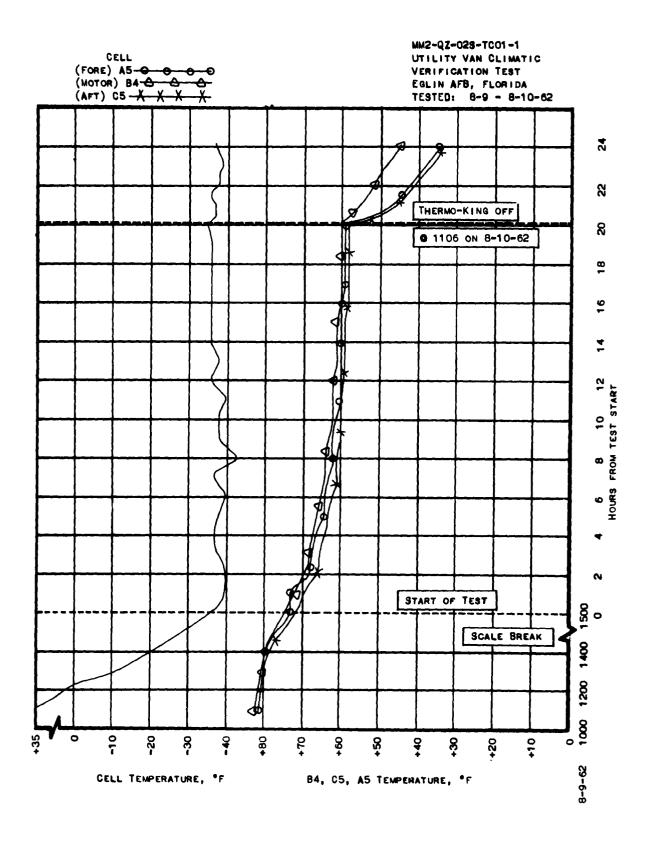


Figure 10



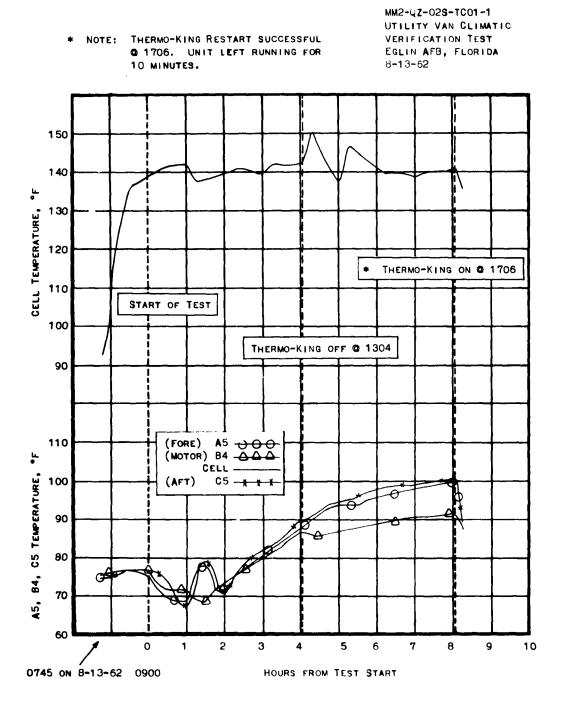
Low-Temperature Test Data (Plot of Thermocouples A2, B6, and C2)

Figure 11

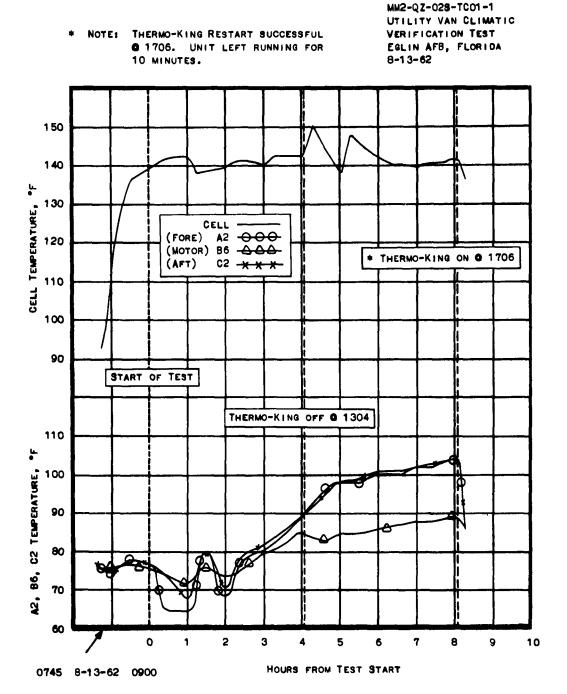


Low-Temperature Test Data (Plot of Thermocouples A5, B4, and C5)

Figure 12



High 'emperature Test Data (Plot of Thermocouples A2, B6, and C2)



High-Temperature Test Data (Plot of Thermocouples A5, B4, and C5)

Figure 14

APPENDIX A

Minuteman Engineering and Inspection Summary



SOUD ROCKET PLANT SACRAMENTO, CALIF		MES NO. DATE HT_ & & 19 July &	946 94 1 of	1 × 1 × 100
MINUTEMAN	PREV. MEIS	S N.A. NEXT MBS		N.A.
THE UTILITY VAN ENVIRONMENTAL VERIFICATION Verification VAN AF 346873 SEIM NO. USAF #3	APPUCA	APLICABLE DWGS:: T-4,92674A APLICABLE SPECS:: Test Plan 752 A		
ENGINEERING INSTRUCTION CHANGES	9 X	INSPECTION SUMMARY CHANGES		INSP. DATE AND STAMP REMARKS
NO ENGINEERING CHANGES		NO ENSPECTION CHANGES	10.000	
APPROVED: PROJECT THE SEQUENCE WHEN THE		APPROVED: INSPECTION MILES	July August Augu	

Page A-1

	1000	3. 3.	TO STATE OF THE PARTY.		06 10
	35	MELS SUC	Table onle		5
SOLID ROCKET PLANT SACRAMENT MITE.		m-624	18 July 1962		
	TITL	TITEUTILITY VACO	inconstruction of the control of the	uo	
ENGINEERING AND INSPECTION SUMMARY		Na Pas	SHRIAL NO S19592	Utility Van S/N 37290	DO.
ENGINEERING INSTRUCTIONS	OP.	INI	INSPECTION SUMMARY	INSP. STAND	REMARKS AND DATE
1. After completion of each step disclosed in the Engineering Instructions of the MEIS, the assigned Test Engineer or Technicians shall apply his initials in the column provided.		INSPECTOR:	INSPECTOR: When a characteristic of this summare		
2. During actual operation, revisions to Testing Requirements contained in the MEIS will be made on the MIPR. Changes will be initiated by the cognisant test engineer. All entries on the MIPR should be made in a legible fashion with black ink. In no case shall the MEIS be altered except for recording the required data as requested.		is not complied wi and record its num column. Accord on complete descritti ancy. This inform the methods and/ors tures, weights, bo stitute materials.	is not complied with, initiate an I.P. and record its number in the remarks column. Record on the I.R. the information that is necessary to give a complete description of the discrepancy. This information should include the nethods and/or the materials used such as pressures, oure times, tempiratures, weights, bonding agents and substitute materials.		
3. When the MEIS is complete the assigned test engineer shall apply his signature in the space provided. The completed MEIS will then be farmarded to the Aerojet-General Quality Control Department.		INDICATION OF INSPECTION 1. Date and Stamp off rel tions on ACC traveler tag.	INDICATION OF INSPECTION ACCEPTANCE 1. Date and Stamp off related operations on ACC traveler tag.	X.A.	
i. Notify Inspection before proceeding with paragraphs marked with a double asterisk (**). Inspection disposition of previous or current operations required.	· · · · · · · · · · · · · · · · · · ·	2. Stamp off e this summary as 3. Forward com ning to Line On	 Stamp off each characteristic of this summary as it is complied with. Forward completed integrated planning to Line One (1) Inspection Office. 	N.A.	
		VOTE: THIS INTEGRATED PLA VALID AND SHALL NOT BY USE TION UNLESS COVER PAGE (M) 053-032, IS AFPROVED BY IN PLAVYING, EACH PAGE OF TH HAVE THE APPROVAL OF INSPE NING AVE DALLITY ENGINEER.	VOTE: THIS INTEGRATED PLAVING IS NOT VALID AND SHALL NOT BY USED RY INSPECTION INVESS COVER PAGE (MIPR) AGG 3-053-032, IS AFPROVED BY INSPECTION PLAVYING. EACH PAGE OF THE MEIS SHALL HAVE THE APPROVAL OF INSPECTION PLANING AND SHALL INGENER.		
			Se sa		

OF TO	PEWARES AND DATE											
PAGB	INSP.											
I SSUE DATE	INSPECTION SUMMARY											Dr. P.
MRIS NO.	INSPE											
101	OP. INIT.											
MINUTEMAN ENGINEERING AND INSPECTION SUMMARY	ENGINEERING INSTRUCTIONS		This Himteman Engineering and Inspection Summary covers the Engineering and Inspection requirements necessary to verify the accept- ability of the 2nd Stage Minuteman Transporter (Utility Van) For operational use under extreme climatic conditions.	References Required	Test Plan 752, AGC Minuteman Division.	Drawing T-4,92674A, Trailer, Transport,	Contract change notification #182 to L. C. SAll to AF 33(600)-36610.	Applicable drawings, sketches, specifications, and procedures in accordance with Page 1 of attached MIPR.	Items to be Tested	AGC 2nd Stage Minutemen Transporter, T-492674. S/N to be specified on the attached MIPR. Installed in the transporter is an inert 2nd Stage Minutemen Motor, 44TE-1, S/N 519592.	K-30 mammass eyele Thermo-King unit with 3-1/2 KW alternator heater assembly. (Assembled to trailer.)	
MINUTE		Scope	ויו	Refe	2,1	2.2	2.3	2°p	Item	3.1	3.2	
2		1.0		2.0					3.0			

OF 10	REMARKS AND DATE						,					
PAGB	INSP. Staie		®				(E)					
I SSUE DATE	INSPECTION SUMMARY		hal Verify thermocouples are located per Figure #1 of Test Flan 752				5.2 Temperature recorder has evidence of valid calibration. method: visual; verify per eticles or dealth					P MES
MRIS NO.	INSP		hal Verify the				5.2 Temperatu of wald					
101	OP. INIT.		•									
MINUTEMAN ENGINEERING AND INSPECTION SUMMARY	ENGINEERING INSTRUCTIONS	Test Equipment Furnished by AGC.	Twenty-six thermocouples installed per Figure 1 of Test Flan 752.	<pre>h.l.l These thermocouples are of Iron-Con- stantan wide and are connected to a terminal strip on the aft end of the van. They are identified per Figure 1 of Test Plan 752.</pre>	Test Equipment to be Furnished by AFGC.	Climatic chamber capable of maintaining temperature extremes of -35°F (± 5°) and +110° (± 5°F), and accommodating a transporter of the following dimensions: 11.25° high, 21.0° long, and 8.0° wigh. (The loaded weight is approximately 23,000 pounds.)	Temperature recording instruments capable of recording 26 channels of data, (Minneapolis-Honeywell Brown Electronic Strip Chart, Y153E62-P16, or equivalent) with recording range of at least50°F to +150°F.	Ducting to went the exhaust gases of the Thermo-King unit to outside air.	Lead-wire, Iron-Constantan, as required to connect each TC to the recording Instrument.	Still photography equipment.	Tractor, with standard 5th wheel, for van movement as necessary	
MINUTER			4.1			5.1	5.5	5.3	4°5	5.5	5.6	
2		0.4			5.0							

4 OF 10	AND DATE				اشسيست								
PAGE 4	INSP.			_									
I SSUB DATE	INSPECTION SUMMARY					Verify free air thermocouples (Wan tc.s) Free from contact with van, and werify motor	t.c. securely attached method: visual						
MRIS NO.	INSE			· ·		6.1.1.1 Veri	• • •						
101 101	OP. IMI.				W V	γ	غ ا	4.15	- i.	+	_	· · · · · · · · · · · · · · · · · · ·	
MINUTEMAN ENGINEERING AND INSPECTION SUMMARY	ENGINEERING INSTRUCTIONS	Gasoline for Thermo-King. (Standard, automo-tive)	Engineering Instructions	Pre-test procedures.	Inspect condition of all internal thermocouples.	6.1.1.1 All free-air TC's must be secured so as to prevent "swinging-in-the breese."	6.1.1.2 All motor-mounted TC's must be firmly comented to the chamber exterior,	Impact the general condition of the motor, harmes, and the-dorms. List any discrepancies found.	Perform functional check of the Thermo-King temperature control unit.	Comment all thermocouple lead wires from recorder(s) to terminal strip.	6.1.4.1 Record the following:		
MAN E	EDG!	Gasols tave)	mering	£	6.1.1			6.1.2	6.1.3	6.1.4			
NUTE		5.7	T A	6.1									
2			0.9										

0F 10	REWALES AND DATE		
PAGB 5	INSP. STANE		
I SSUB DATE	INSPECTION SUMMARY		Mr mass
MRIS NO.	INSPE		
الالك 101	OP. INIT.		
AND INSPECTION	TIONS	G FEET OF WANTALISM F WINTERSOF	
MINUTEMAN ENGINEERING A	ENGINEERING INSTRUCTIONS	Channel (Tarie) (1 w 4 200 200 200 200 200 200 200 200 200 2	
MINUTEMAN	EN	174 Becorder 174 184 174 175	

OF 10	AND DATE				
PAGE	INSP.	6	(()	(a)	
ISSUR DATE	INSPECTION SUMMARY	6.1.5 Verify still photos taken of internal instr. and test setup. Wp. Method: visual	֓֞֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓	Varify cell temp is at -35°F ± 5°F (this is start of test) Record time and date. Method: visual	<u>o</u> c
MBIS NO.	INSPE	6.1.5 Verify stinternal up.	6.2.3 Verify Record Record	Date 6.2.3.1 Verify 4 5 T (t Record t Method:	Time Date
ا01 101	OP. INIT.	dist	/ (
MINUTEMAN ENGINEERING AND INSPECTION SUMMARY	ENSINEERING INSTRUCTIONS	6.1.4.2 List recorder speed: der No. 6.1.4.3 All data channels che satisfactory operatic Stifl photos taken of internal mentation, recording system au general test set-up. Verify full fhel supply on The King. (26.706. 67647 6.00 Merify steps 6.1 through 6.1.6 plated. Thermo-King turned ON and set tain +80°F (.5°) within the ve (Note: Use Thermo-King temper indicator as the reference.)	a ti	0.2.3.1 Mecord actual time -35 (15 P) is reached. This time shall be considered at the start of the test.	Time (500 Hours 940762 Date

Page A-7

			····					
OP 10	REMARES AND DATE			į	Morres Temp, PASSF. AVE. VAN TEMP TEMP COCD - 574	ATTENDY ONSWERSE		
PAGE	INSP. STANE	(k)			@			
ISSUB DATE	INSPECTION SUMMARY	6.2.4 Cell temp. maintained at -35°F ±5°F for four hours after temp. equalibrium is reached on the motor surface. Method: visual			6.2.5 After completion of 6.2.4 above Thermo-King shut down and cell temp, maintained at -35°°° ±5°°°°°°°°°°°°°°°°°°°°°°°°°°°°°°°°			M WEED
MBIS NO.	INSPE	6.2.4 Cell to # 5°F ; equal 1) motor Ma			6.2.5 After of Thermotemps of Corford Park			
101	OP. INIT.	<i>y</i> 0	* 3	A.	R	MA	S. F.	
MINUTEMAN ENGINEERING AND INSPECTION SUMMARY	ENGINEERING INSTRUCTIONS	6.2.4 The low temperature environment will be maintained until temperature equalibrium is reached on the motor surface, (Use TC Bk as the reference), and continued for an additional four (b) hours.	6.2.4.1 Record stabilisation time. TO Bi 0700 Time 10 Access Date 6.2.4.2 Record stabilisation temperature. Temp. +60	During the enti be read and man hour. (To inc) chamber therm	6.2.5 At the completion of 6.2.4, shut down Thermo-King, and maintain chamber temperature at -35° (±5°) for an additional four (\(\bar{\eta}\)) hours. Continue to monitor and record all TC date. Record time test completed.	1/06 The 10 40 f6 2. Date 6.2.6 Glimatic chamber temperature returned to ambient. 1830	6.3 High Temperature Test 6.3.1 Verify full fuel supply on Thermo- King.	* (5.57 - 1.00 Succession)

8 .OF 10	AND DATE	(s)		رو مہ کر و	5 0 \$1, \$ cv13	מוס-יבומל מחיי החוד ימכא ביי	1460 THER	' ' 'E'9
PAGE	INSP. Staie		C,		(
ISSUE DATE	INSPECTION SUMMARY		6.3.3 Verify van temp. is +80°F + 5°F and cell temp. is +110°F ± 5°F and record. (This is considered start of test) Van temp. at 80° 8/7 date o' time cell temp. at +110° 8/2 date	Method: visual	6.3.4 The cell +140° environment maintaine tained for four hours during which time the Thermo-King is running. Hethod: visual Connected (300 Mrs.	6.3.5 After completion of 6.3.4 above Thermo-King unit shut off and cell temp. maintained at +110°F this environment maintained for four hours. Hethod: visual		A W. A
MRIS NO.	INSPE		6.3.3 Verify and ce and re start Van tee	£	6.3.4 The cell tained function till running. Meth	6.3.5 After compl Thermo-King cell temp. this enviro four hours.		
101	OP. IMIT.	X	· ·		8		<u>.</u>	
ENGINEERING AND INSPECTION SUMMARY	ENGINEERING INSTRUCTIONS	Turn on Thermo-King unit and adjust for 80°F (± 5°) within the wan.	When van interior stabilises at +80°F (± 5°), increase climatic-chamber temperature to +1µ0°F(± 5°). Record actual time the chamber reaches +1µ0°F (± 5°). This time shall be considered as test-start-time.	0500 The 8-13-41 Date	The +140° environment shall be maintained for four (4) hours with the Thermo-King running. All temperatures shall be read and mammally logged once each 15 minutes. Highest temperature recorded on the motor amfines: \$\empsycolor{86}\$ \text{***} ***********************************	At the conclusion of the first four hour period, the Thermo-King shall be turned off and the chamber temperature continued at +110°F for four (4) additional hours. Continue all data readings during this four (4) hour period.	At the conclusion of the second four (h) hour period, the climatic-chamber temperature may be reduced to ambient, and the floraciting temperature for the floraciting temperature.	
MINUTEMAN ENGINEERING SUMMARY	ENGI	6.3.2	6.3.3	•	4.3.4	6.3.5	6.3.6	

PAGE y OF 10	REMARES AND DATE										
PAGB	INSP.		0								Δ.
ISSUE DATE	INSPECTION SUPMARY	Hentify all data and records		Method: Indicate at the end of all records and data						7.0 Indicate by signature and title that the test was preformed per this MEIs and MIFR.	Sylver Red Hack & Kenter
MRIS NO.	INSPE	6.4 Edentigy	objective	79						7.0 Indicate that the this MEI	i O
008 101	OP. INIT.	100	X	<u> </u>	*	<u>;</u>	A.	_			
MINUTEMAN ENGINEERING AND INSPECTION SUMMARY	ENGINEERING INSTRUCTIONS	6.h Post-Test Operations	6.4.1 Verify validity of all data.	6.4.1.1 All records identified. (Record number, date, TC number vs channel assignment, etc.)	6.4.2 External instrumentation cabling removed.	6.4.3 All entates completed on this MEIS.	6.4.4 Transportation Officer notified to arrange ahipment of Utility Van Motor to:	Aerojet-General Corporation Solid Bocket Flant Building 4915 Mimbus, California Attention: T. J. Harrell	6.4.5 The completed MEES, thermocouple recordings and any other test data or reports are to be given to the witnessing representative from AGC.	7.0 I certify that the testing of the transporter, the Rocket Motor, was accomplished par the requirements set forth in this MEIS and Test Flan 752.	Stened a Conference of Confere

	f ,]		
,v	10 OF 10	REMARKS AND DATE	
AGC 3-052	PAGE	INSP. STAN	r: X
ÀG	ISSUB DATE	INSPECTION SUMMAY	L. W. Oll graves
	MBIS NO.	INSPE	My Plaining My Ling My Lendent Ling Brotineer
	101 101	oe. IMT.	
	MINUTEMAN ENGINEERING AND INSPECTION SUMMARY	ENGINEERING INSTRUCTIONS	Written by: J. D. Soll Environmental Program Engineer Environmental Testing Department Solid Bocket Flant

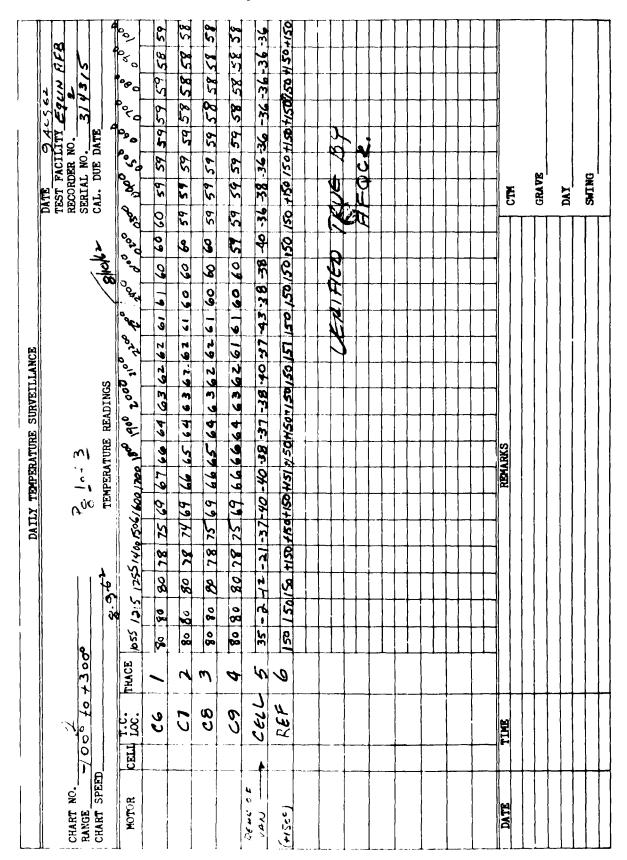
APPENDIX B

Temperature Surveillance Chart,
High- and Low-Temperature Tests of
Aerojet Utility Van



CHART NO. RANGE CHART SPEED	1000/-	7		3000		7		200	-0 -	LOT 3		SANTA	SURVETLLANCE				311	ي	DATE TEST RECO SERI CAL.		FACILITY DER NO. L NO. DUE DATE	32 650	2007 2007 2007 2007	
MOTOR	CELL TO		(9)	1,520		2	- 11 - 11	1 3	1600 1700	2/00	06 08 °	P2	00-11	6,3	38.7	OOK 1	8		Oak O	O STO	0090 0	0.00	988	000 000
		7 4 8 6	- 2) m4	ಇಇಇ:	واسالا	0200	: _ 	-+-+-+	1-1-		499			4 20	4000	222	272	3 5 3	0-0	9 09 09	09 0	500	609	58 58 60 60
		1001	100		+ + + 2	226	11-1-	7227	000	1 1 1	-1 - 1 - 1	<u> </u>		634	250			3 2 3 3			590	2007	5 55 65	58 58 58 58
Figorst		000	000	8 8	300	787	780		99 69	650		663	400	0 4 0 - 4 0	300	0000	500	2528	• • -	59 59	00 to	20 00 P	382	58 58
((447)	2 2 2	3/2	0/1	¥≊3	افسا ایدا	80	78 .			500	0	563		-32	5-38	404			4 20 2	1 1 1	200	200	-∔-┺∔ `∔ `	4474
	200	883 7 7 7	です なべ			150/		╶ ╀╼╂ _╼ ╂┈	1 T T		_	2000	62 6 4 5 65 67 6 6 65 67 6 6 65	49	500	500		500			# ++ 5+		200	60 60 60 60 4/50-/52
		200	1,000	200		288	+-+*-	+-+-	99 69	+ + - + -	900	200	0490	250	99	3073	35 53 C	2 36	2000 2000	12 4°	3000	300 23	32 25	00 00 00 00 00 00 00 00 00 00 00 00 00
		7450				80 7 80 2 8/8	79 7 80 7		 		353	926	1 1	429	4 -4 -4	1-1-1-		 		900		1 1 1		5000
		0.0 0.0	R _R	≅ ≅	0 0	000	78 7	9	99	66	70	63	79	 				30		000	+			
77.4			1/194								Ш						3		2	80		11	Q	V
9			CAUS INTE			3, 8		77 71 8		Clywood Cos - Lextus Loveywell	\Q\ \	9 43	9 7	10 KE	1 1811	C KEF JUNCTION POTENCY CF. 2.	3 1		GRAVE	_ &				
																			DAY	<u> </u>				

Page B-1

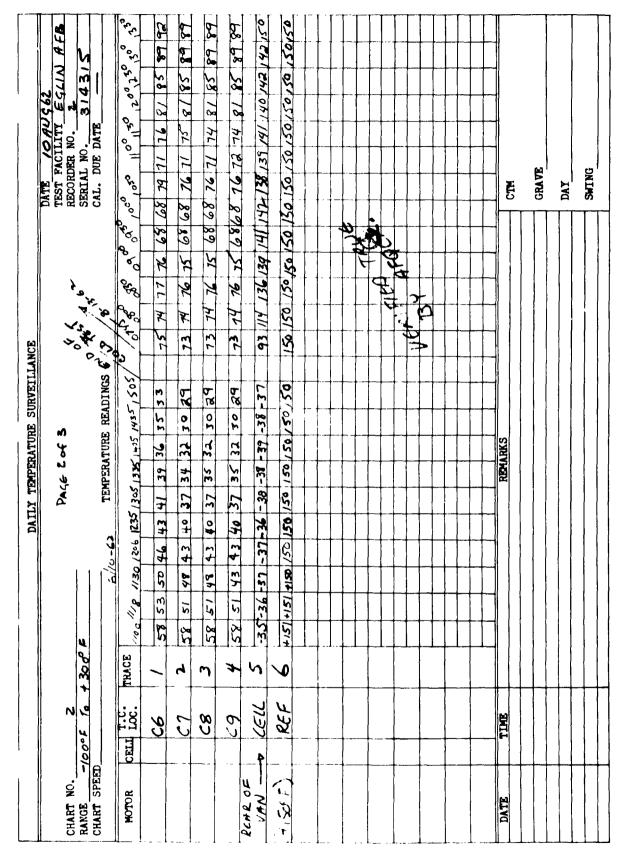


Page B-2

- No.								2	Page	3	mi		Page 2203				2		TE	DATE 10 4 US TEST FACILITY E	CILLI	17 E	292 275	62 ALP	80
RANGE CHART SPEED	100°1	0 / m.	+30	000			458	7	TEMP	TEMPERATURE READINGS	32 32	EADII	NGS 6	Test Series S		is lie	2		S S	SERIAL NO.	MO E	101	22.0	90	
MOTOR	CELL	1.C 1.OC	TRACE	00	00	8/	3	11 V	505	1585	5 405 436	305, 28	se S	0		0.00	OBO		80	050	// A	1,30	1002	230 (3	0661 0061
	_	14	, 4	38	10 17	25	99	2	2 3	44 40	2 2 2			7.2	15 7	85 %	11	69	69		72 7	79 8	76 /	-	95
		2 8	m	200	200	2	* +	+-+-	+-+-	+	++-	 	0=	177		\$ 2	12	60	_		+ +	+-+-	1.1.		<u>ا</u> ـا.
		£;	5	00	24	तेष	+-+	1	+.+	+++	+-+			42,	45/	7	25	22	20	-++	+-+	1	10	7. T.	+ +
		200	9	10 or	5 3 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4	20	47	777	7 5	38 38	3 36	4 2	> ~	2 3	5,5	700	76	2/2	5 5	100		1 2 0 0		20 70	- +.
	_	48	00	900		_	+-+	+-+	+ +	+	+-+	+ +	(<u> </u>	7	3/2	11	16	1 1	20		7 -7				200
Fear of	L	43	6			20						× 1	2	H	7	290		11	11			17	-	1	58 8
NAN	-	CELL	6	22			•		_		M V	2000	3 .	21.	ह ५	377	32	ž.	13,		<u>≂</u> †−	140/4/	4	2] (Ja
		4	7				3 6	2 %	5 4	53 51	+	1		7		180	-	200	300	2/3	A LA		- 1	200 200	2 0
		3	57	-	56	54			1-1	11	+	1	-	77	4	81	11	89	89	8	† 1	+-	82	858	705
		8	4	9	5.6	28	\vdash	-	\vdash			45		16	1	11	11	12			- 71	7780		83 8	
		REF		450+1504150	• 150	25,4	-7			-+			۱ و	8		3	150	120	-			15'0/50		_	
	+	0 0	٥	2	3		-4-	+-	+	52.50	10	0 0		• 7	9 7	7	2	77	1	_		9	١,		20
		8	90	7 9	10		3 5	23	2 2	+	+-			1,2	7,	5 %	2 %	1	5 6	\$ 5	15.	900	00 83	—	00
		66	Ø	-	53	+-	+	╅┈	╃	+	_			3	•	2	2	1	2				↓	58 56	1-
		7	8	59	55		Н			42 40	85 0	_	7	11	10	18	11	26	89			198			1
		3	7		35		-+		44 #	43 40	_	_		7	×	18	11	74	99	8	7/8	_	-+	\neg	
	1	3	11	_	55	-	+	-+		+	m	m	,	7	15	28	7	H	200	_	-+	\rightarrow	78		তা
		3,	23		W.	+	9	43	٦,	37 %	n i	<u>ا ا</u>	~ ^	4	12 x	>	2	27	*		10	2000	A C	χο <mark>υ</mark>	1 22
		3			1	*	1-1	+-+		╂╌┥	1	4-1	H	3	4-4	†		1	9	$\ddagger \dagger$			3	1-1	}-
	1					1	+	\dagger	+	+	-	\downarrow	\downarrow				7	1	1	+	+	+	+	+	+
							1	\dagger	+	+	+	+	-	\perp		3	\ \frac{1}{2}		1	T	+-	╁	+	+	+
DATE		110							И	REMARKS	S					-		1	ŀ	į		1	1	╢	1
0//8	77	2011	1116	THERMO-KING	12	12	12	TURNED	1 1	OFF	J	9/1/02	9		17	 			TT	E S					
8/13	0 745	75	STORT	AT	0	1 !	HIGH	1 1	dw3/	1 1	7557			1					у ТТ	GRAVE					
ļ	0000	8	BOTH CELL	Y H	CELL	- 1	707	2	14-5PEC	ان			-	-	ؠؙٟۮ				7	ļ					
	130	9	176	247	4	1	3	280080	. [940			7						<u>-</u>	DAX	1			-	-
	/ 700	0	7257	.	2000	JI.	21.5						7	1					T	CALL					

AGC 3-01-7 500-1

Page B-3



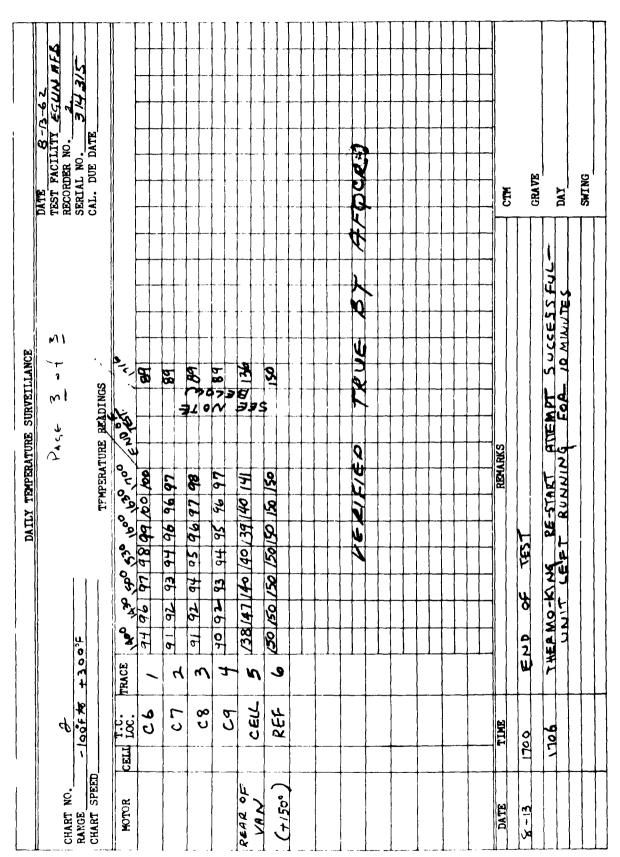
AGC 3-047.500-1

Page B-4

TEST FACILITY ESC., RFG. RECORDER NO. / SERIAL NO. SERIAL NO. SERIAL DUE DATE GRAVE SMING E DAY コマウー Success fur DAILY TEMPERATURE SURVEILLANCE TEMPERATURE READINGS 11006 OF THERMO-KING Page REMARKS 10 MINUTES ATTENPT ON FOR TEST 20 88789 13889 13989 13989 13989 END OF PE-START **FEF** -bo + + + 300°F TRACE REF REF 9 K 96 88 4 CELL ΰ **C3** 15.5° P £ 9 8 8 87 CS 1200 1206 CELL FROM OF VAN 180 CHART SPEED CHART NO. HOTOR DATE -13

AGC 3-047-600-1

Page B-5



4GC 3-047-600-1

Page B-6

APPENDIX C

Chronological List of Events,

High- and Low-Temperature Tests of

Aerojet Utility Van



Aerojet	1013 Rev -General tion, SRP	MOTOR LOG OF TIME VS OPERATION W.C. NO. 0625-26-014
AGC 3-12	28-631	ENVIRONMENTAL VERIFICATION - CLIMATIC TYPE OF TEST
C.R. NO	. N.	A. TEST ENGINEER J. D. Sohl
PROGRAM	MINUTEM	AN 2ND STAGE TEST SERIES MM2-QZ-02S-TCO1 SHEET 1 OF 4
		AGC Utility Van, VR-1152A. S/N 37290; AF 348873
DATE	TIME	OPERAT ION
Sacto		
7-24-62	1420	Utility van arrives at Bldg. 4605 for thermocouple installation.
		Instructions issued to Instr. Foreman. Work initiated. Completion
		estimated by 2300 hrs.
7-25-62	0800	Instrumentation continues - approx. 6 hrs. work remain.
	1430	Instrumentation completed (per Test Plan 752A).
	1530	Van removed to 4615 pending shipment to Eglin AFB.
7-26-62	1600	Van departed for Eglin AFB. Carrier is Leonard Bros Shipper # S35876.
0-62		Van arrived O.K. at Eglin AFB. Notified by J. Cole (AGC - Eglin) that
		there will be a four (4) day delay getting into the climatic chamber
		due to chamber schedule blippage,
Eglin	AFB	
8-8-62	0900	Inspected van and interior instrumentation. Everything in satisfactory
		condition.
	0930	Instrumentation group informed of test requirements. They will provide
		(2) Brown I-C 24 ch. recorders, and make hook-up to same. Unable to
		Move van into chamber until late afternoon due to previous test in
		progress
	1100	Ruquested inspection coverage prior to start of test. Coordination
		meeting set up for 1300 hrs.
	1545	End of day shift - lab cleared of all p. vious test equipment. Van
L		brought to lab area. Calibration of (2) Brown recorders complete.
8-9-62	0730	Van brought into climatic lab.
	0830	Thermoking crankcase oil changed. (10W-30) put in.
	1000	Instrumentation complete - photos taken. Thermo-King set to +80°
		and running O.K.
	1030	Pull-down of climatic chamber to -35° begun.
	1055	Temp. of motor (B-L) = 82°F.
		Temp. of cell (aft) = +35°F.
	1500	Cell temp. (fore)30°. Official start of test.
		C-1

ST Form	1013 Rev	2 TEST OPERATIONS PAGE
	General	MOTOR LOG OF TIME VS OPERATION W.C. NO. 0625-26-014
or por a	20, 0	ENVIRONMENTAL VERIFICATION - CLIMATIC
AGC 3-12	8-631	TYPE OF TEST
C.R. NO	N.A	• TEST ENGINEER J. D. Sohl
PROGRAM	MINUTEMA	N 2ND STAGE TEST SERIES MM2-QZ-02S-TC01 SHEET 2 OF 4
TEST SPI	ECIMEN _	AGC Utility Van, VR-1152A. S/N 37290; AF 348873
DATE	T IME	OPERAT ION
8-9-62	1502	Noticed intermittent operation of Thermo-King. Trouble is somewhere
		in switch-box.
	1505	Refueled unit.
	1519	Thermo-King back on - still intermittent operation. Lost 10 degrees
		in van.
	1500 -	Continuous intermittent operation of the unit. Trouble shooting the system
	104)	disclosed faulty fuse holder/fuse. Repaired same.
	1900	System working O.K. Refueled Thermo-King. Excess frost is noted forming
		on the cendenser face (Thermo-King).
	1915	An excessive amount of oil was found on the top side of the engine
		power section. Unable to determine source of leak.
	2230	T.C. A-9 indicates 59+° - cell temp. @ van front is -31°. Accumulation
		of oil on engine increased; approx. 1/3 of condenser face is frost.
8-10-62	0045	Re-fueled unit - heat off for approx. 5 minutes.
	0250	Inspected van and Thermo-King. Only apparent trouble is the afore
		mentioned oil leak and accumulation on top of engine. Oil level
		checked O.K.
	0608	Re-fueled unit. Heat off for four (4) minutes. Oil level checked
		O.K. Still evidence of leakage. Fuel gauge is in-op.
	1025	Checked voltage across heater fuses; 208 vt each leg. Oil dripping
		from power unit.
	1106	Shut down Thermo-King. Oil level O.K.
		This completes the test of the Thermo-King operation - four (4) hours
		of temp. stabilization on TC B4.
** *		In Summary:
		Van interior at +80°F ⊌ 1030 hrs. on 8-9-62.
		Start pull down of cell 3 1030 hrs. on 8-9-62.
		Cell in spec at -35° (±5°) \Rightarrow 1500 hrs. on 8-9-62.
		Motor temp. stable (B-4) = 60° 0700 hrs. on 8-10-62.
		Thermo-King off (Stab. plus 4 hrs.) 1105 hrs. on 8-10-62.
		Test complete - 1505 hrs. on 8-10-62.

Aerojet	1013 Rev General tion, SRP	2 TEST OPERATIONS PAGE W.O. NO. 0625-26-014 ENVIRONMENTAL VERIFICATION - CLIMATIC
AGC 3-12	8-631	TYPE OF TEST
C.R. NO	N.A.	TEST ENGINEER J. D. Sohl
PROGRAM	MINUTEMA	N 2ND STAGE TEST SERIES MM2-QZ-02S-TCOL SHEET 3 OF 4
TEST SP	ECTMEN	AGC Utility Van, VR-1152A. S/N 37290; AF 348873
DATE	TIME	OPERAT ION
		Final motor temp. (B-4) +45°F.
		Average van temp. +35°F.
		Thermo-King re-start attempt: Unsuccessful.
		Low-temp, warning light was on below 70°F.
8-13-62	06112	High temperature (+lh0°) test scheduled for today. Climatic chamber
		maintained at +80° over the weekend.
	0700	Thermo-King serviced with SAE 20W-40 oil and refueled.
	0745	Motor and van at 80° (±5°) - chamber set for +140°. (Reads 95° @ 0747)
<u> </u>	0900	Official start of four (4) hour test at +140° (15°) with Thermo-King on.
		Cell reading: 135° Forward
		141° Aft
		As all interior TC's are reading approx. 75°, I changed thermostat setting
	0915	to +75° from +80° to prevent unit from going on heating cycle. Refrig. cycle manually turned on by again reducing thermostat setting -
	0919	had to be certain system was still operational. All TC's showed rapid
	0930	drop to around 70°. Reset to +75° on Thermo-stat. All temps. still around +70°.
	1010	Reset to +78° to let van warm up. Temps. \(\omega 68-70^\circ.\)
	1230	Temps. within van show steady rise (most around 84-86°). Thermo-King
) 		still running normally - although the suction pressure in the evaporator
		has risen from around 40 psi w +75° to 68 psi w 88°. (Thermostat is
		still set w 78°.)
	1300	End of four (4) hour period of Thermo-King operation.
		Highest temp. on motor (B-2) is 86°.
		Average van temperature is 88°.
		Also noted that oil leak which was apparent during cold-temp test is
		not present. Gasoline gage hasn't worked throughout entire cold or
		hot tests.
	1305	Thermo-King turned off.
	1405	Re-started Thermo-King and ran for 1 min. to obtain head pressure
		reading as requested by STL Rep.

Aerojet	1013 Rev -General tion, SRF	MOTOR LOG OF TIME VS OPERATION
C.R. NO	N.A.	TEST ENGINEER J. D. Sohl
PROGRAM	MINUTEM	AN 2ND STAGE TEST SERIES MM2-QZ-02S-TC01 SHEET 4 OF 4
TEST SP	ECIMEN _	AGC Utility Van, VR-1152A. S/N 37290; AF 348873
DATE	TIME	OFERATION
8-13-62	1405	Van temperature - 95°F.
0-19-02		Head pressure - 260 psi.
		Suction pressure - 40 psi.
		(Note that this short cooling period reduced most interior temps approx.
		4 degrees.) Overheat warning light operation normal as interior temps
		reached 90°F.
	1705	Test complete. Cell will be returned to an ambient condition. After
		four (4) hours in the +140° environment, (without Thermo-King cooling).
		the follow ng temperatures were recorded:
		Highest temperature on motor: plus 94°F.
		Highest temperature in van: plus 104°F.
		Average temperature in van: plus 100°F.
	1706	
	1100	Thermo-King turned on - operating normally, No further evidence of oil leak.
		Head press. 270 psi. Van temp. 102°F.
		Suction press. 45 psi.
	1711	After five (5) minutes of Thermo-King operation, the average van
		temperature was reduced to 92°F. (Cell still +136°.)
	1716	After ten (10) minutes of Thermo-King operation, the van temperature
	1/10	(average) is down to 90°.
		(2.01280) 15 4041 00 /0 8
	<u> </u>	
_		
-		
-		

UNCLASSIFIED 1. High, and Low-Temperature Tests of the Aerojet-General Utility Van 1. Aerojet-General Corporation 11. Aerojet-General Corporation 12. Aerojet-General Corporation 13. Force Ballant Systems Division, Arr Force Systems Command, USAF 111. Contract No. AF 33(600), 26610 UNCLASSIFIED	UNCLASSIFIED 1. High, and Low-Temperature Tests of the Aerojet-General Unity Van 1. Aerojet-General Corporation II. Arr Force Ballistus Systems Divason, Air Force Systems Command, USAF III. Contract No. AF 33(600)-36610 UNCLASSIFIED
Sacramon, Californa Marojet General Corporation Sacramon, Californa HIGH-AND LOW-TEMPERATURES TESTS OF THE ALROET GENERAL UTILITY VAN, by J. D. Sohl Technical Roport, Splember 1962 6 Pp. AF BED 'TH-88D-TDR-92.331 Uselsanited Report The Acrojet-General Utility van successfully completed high and how-temperature test to verify the accepta- bility of the 'van as a treangort vehicle of Minutena second-stage operational motors. An analysis of the data obtained in the tests is presented.	Astroyat-General Corporation Sacramento, California HIGH-AND LOW-TEMPERATURES TESTS OF THE ARROLTS—CEMERAL UTILITY VAN. by J. D. Sahl Technical Report. September 1962 6 PP. Is line. ARSO-TH-TDR. 62-331 Unclassified Report The Astrojet-General Utility was successfully completed high and for-temperature tests to verify the accepta hilty of the van as a transport whele for Mindsonna second-stage operational motors. As analysis of the data obtained in the tests is presented.
UNCLASSIFIED 1. High and Low-Temperature Tests of the Aeropet-General Utility Van 1. Ary Force Bullatte Systems Drystock Ary Force Systems Command. USAF III. Contract No. AF 346001-36610 UNCLASSIFIED	UNCLASSIFIED 1. High and Low Temperature Tests of the Aerojet-General Unity Van 1. Aerojet-General Copration 11. Ar Force Ballisti Systems Division. Air Force Systems Command. USAF III. Contract No. AF 316001-36610 UNCLASSIFIED
Acrojet General Corporation Secretary Secretary High-amoutor allocations ALROJIT CENERAL UTILITY VAN. by 1. D. Schl 1. Ceneral September 1942. 6 pp. 1. Technical Proposition September 1942. 6 pp. 1. Technical Proposition Secretary ALROJIT SECTION 64-131 Unicaentied Report The Acrojet-General Utility was necessfully completed Nath and low-temperature tests to verify the acceptability of the van as a ranaported. For Minuteman second stage operational meters, An analysis of the dath obtained in the tests is presented.	Arroper General Corporation Secremento. Californas High AND LOW - TRAPER ATURES TESTS OF THE ARGUA - GENERAL UTILITY VAN. by J. D. Schille III (1912). Testourist Report. September 1962. e. pp. 18 Illia. J. Appendises APBAD - TN. 850 - TN. 62-131 Unclassitied Report. The Arroper-General Unity van successfully completed high. and low - temperature tests to verify the accept- billy of the van as a ranaport which for Ministernas second - stage operational motors. A analysis of the data obtained in the tests is presented.

